German Aerospace Center (DLR)

Contributions to 6th EUCASS and to European Aerospace Research





DLR German Aerospace Center





- Research Institution
- Space Agency
- Project Management Agency

Locations and employees

Approx. 8000 employees across 33 institutes and facilities at 16 sites.

Offices in Brussels, Paris, Tokyo and Washington.





A. Herbertz et al

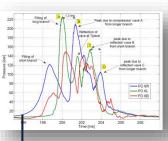
Trajectory Optimization for a Low Cost Lunar Technology Demonstrator Mission

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PP and FD papers

C. Bombardieri et al

Filling process in spacecraft feedlines

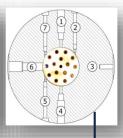




M. Börner et al

Laser Re-Ignition of a Cryogenic Multi-Injector Rocket Engine

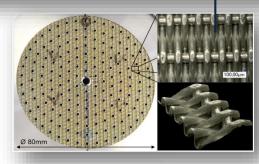




Mission Analysis

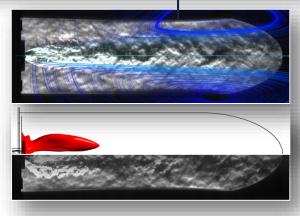
Fluid Flow & Propellant Injection

Ignition – Numerical Simulation & Laser



J. Deeken et al

Axial Pressure and Wall Heat Flux Distribution of a Porous Injector Head (API)



M. Wohlhüter et al

Numerical Analysis of Methane/Oxygen Laser Ignition in a subscale Combustion Chamber



PP and SM papers

Schmierer et al

Project status of the HyEnD STERN hybrid sounding rocket project at Stuttgart



S. Webster et al Asymmetric frequency response of a LOX/H2combustor in frequency

ramping tests

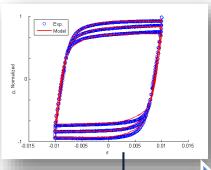
Kobald et al

Development and test campaign of a 10000 N flight weight hybrid rocket engine



Bouajila et al

Modelling of the mechanical behavior of a copper alloy using Chaboche model



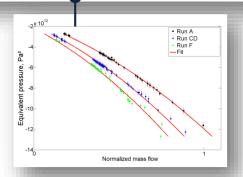
Combustion instabilities

Hybrid Engines



Petrarolo et al

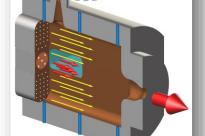
Performance characterization regarding fuel composition and combustion stability of advanced hybrid rocket engines



Materials characterization

Bouajila et al

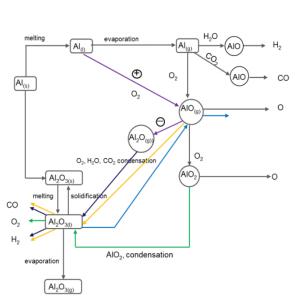
Investigation of the influence of temperature and stress on the permeability of porous continuous fiber-reinforced composite materials through experiment and simulation

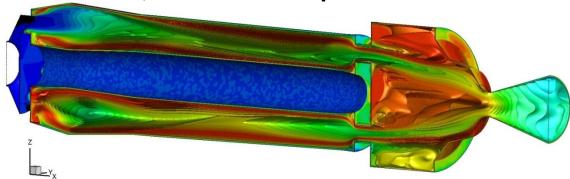


S. Beinke et al

Modelling of a co-axial LOX/GH2 injection element under high frequency acoustic disturbances

Numerical Simulation of hybrid rocket engine combustion chamber processes, aluminum particle combustion, and Green Propellants





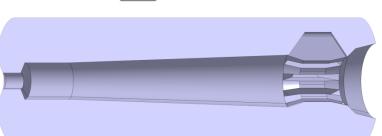
S. May: **CFD Simulation of Chemical Non-Equilibrium Reacting Flow within the AHRES Hybrid Rocket Engine**; PP-Hybrid Prop Modeling

O. Božić: Modeling of the Transformation Kinetics of Small Metal Particles during Combustion inside the Chamber of Hybrid Rocket Engines; PP-Hybrid Prop Modeling

H. Ciezki et al: Influence of various aspects on the performance characteristics of gel populsion systems
PP – Gelled Propellants

M. Negri: Replacement of Hydrazine: Overview and Very First Results of the EC-H2020 Project Rheform PP – Green Propellants

G. Poppe: Optimization of Finocyl Grain Geometries of Solid Rocket Boosters PP – Solid Propulsion I

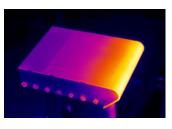


Hypersonics I



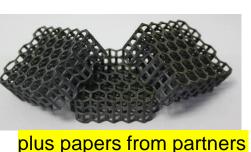
- S. Willems et al.: Free transition on a slender cone in a quiet and a conventional wind tunnel and the effect of ultrasonically absorptive materials, FP
- DLR Cologne H2K conventional wind tunnel vs. Purdue BAM6QT quiet wind tunnel

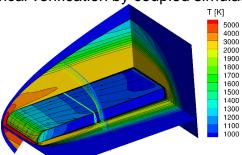
Overview of EU FP7 research project THOR



THORDAN Representations of future space vehi

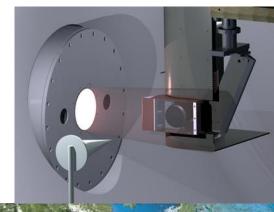
- B. Esser et al.: Innovative Thermal Management Concepts Space Vehicles, Paper 347, SM: Design and Optimization (II)
- Thermal management concepts for highly loaded components of thermal protection systems
- Novel CMC materials and structures
- Experimental verification in high enthalpy ground test facilities
- Numerical verification by coupled simulations



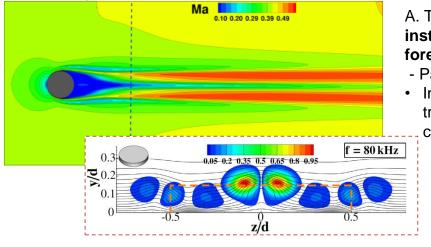


Lars Steffens, Uwe Koch, Burkard Esser, Ali Gülhan: Characterization of Weakly Ionized Argon Flows for Radio Blackout Mitigation Experiments

- Characterization of high enthalpy argon flow states for ground based radio communication blackout experiments
- Experimental simulation of radio communication blackout in L2K facility
 - Experimental verification of ExB Blackout Mitigation Scheme

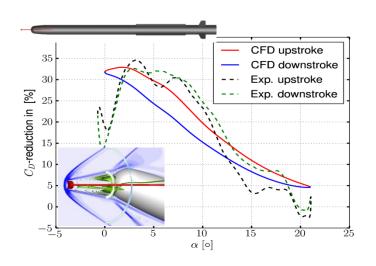


Hypersonics II



A. Theiß and S. Hein: Investigation on the wake flow instability behind isolated roughness elements on the forebody of a blunt generic re-entry capsule

- Paper FP-514
- Investigation on the laminar-turbulent boundary layer transition behind a roughness element on an Apollo-like capsule at hypersonic flow condtions



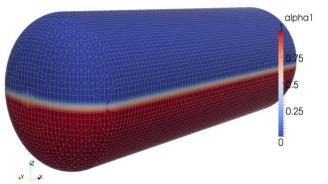
C. Schnepf, E. Schülein: Wave drag reduction with a self-aligning aerodisk on a missile configuration

- Paper FP-470
- Experimental and numerical simulations show the potential of a selfaligning aerodisk to reduce the wave drag on a pitching missile also at high angle of attack.

FP7 funded project on advanced tank technologies for hypersonic







plus papers from partners

M. Sippel et al: **Final Results of Advanced Cryo-Tanks Research Project CHATT**; SM-427 paper

M. Stief, J. Gerstmann: **EXPERIMENTAL AND NUMERICAL INVESTIGATION OF AXIAL AND LATERAL SLOSHING INSIDE A LARGE CYLINDRICAL TANK**; PP - Upper Stages

....

V. Clark: **Modelling of Propellant Management Systems in Early-Phase Launcher Development**; SI - Space Propulsion MDO

A. Hauschild: **Results of the GNSS Receiver Experiment OCAM- G on Ariane-5 flight VA 219**, session FD: FLIGHT DYNAMICS/GNC and AVIONICS for Aeronautic and Space Applications

- First flight of a GNSS receiver experiment on Ariane-5 launcher VA219 with ATV-5 on July 29, 2014, from Kourou into a LEO orbit
- Successful tracking of GPS, Galileo and GLONASS satellites
- Trajectory determination of the launcher's upper stage from lift-off until ATV separation

V. Clark et al: **Process Chain Development for Iterative, Concurrent Design of Advanced Space Transportation Systems**; SI - Space Systems MDO





What about the Future?

